

Power System Analysis

- The goal of the course is to provide an overview of interconnected power system operation.
- Modern methods of power system analysis and design.
- The course will equip the student with the basic tools for analyzing the operation of a power system in normal as well as emergency conditions.

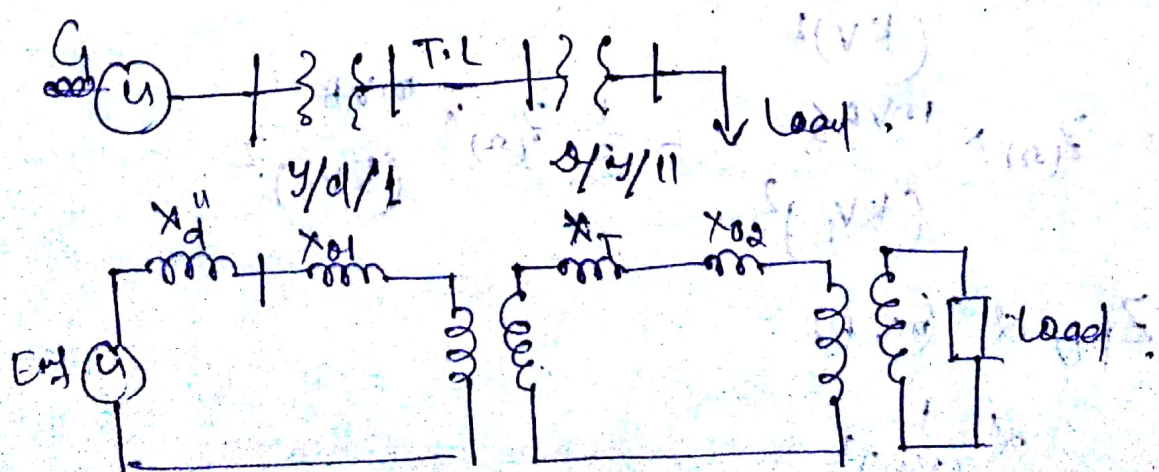
PS-I

— 0 —

PS-II

- | | |
|-------------------------------------|--------------|
| ① Transmission line parameter. | → fault. |
| ② Performance of transmission line. | → stability. |
| ③ Static power equation. | → load flow. |
| ④ Voltage control. | |
| ⑤ power factor correction. | |
| ⑥ distribution system. | |

Single line diagram



To represent the total system in per unit system.

$$\text{Per unit} = \frac{\text{Actual value}}{\text{Base Value}}$$

$$Z_{pu} = \frac{Z(\Omega)}{Z_b} \quad Z_b = \frac{V_b}{I_b} = \frac{V_{rated} \times \sqrt{3}}{I_{rated} \times \sqrt{3}}$$

$$= \frac{V_{rated}^2}{VA}$$

$$Z_{pu} = \frac{Z(\Omega)}{V_{rated}^2} \times (VA)$$

$$\frac{Z(\Omega) \times MVA}{(KV)^2}$$

No unit

* Per unit value is defined in Per-Phase only.

Y-connection

$$Z_{pu} = Z(\Omega) \times \frac{MVA}{(KV)^2}$$

$$Z_{pu} = Z(\Omega) \times \frac{90 \text{ mVA}/3}{\left(\frac{KV_L}{\sqrt{3}}\right)^2} = Z(\Omega) \times \frac{90 \text{ mVA}}{(KV_L)^2}$$

Δ -connection

$$Z_{pu} = Z(\Omega) \times \frac{MVA}{(KV)^2}$$

$$= Z(\Omega) \times \frac{MVA/3}{(KV_L)^2} = \frac{1}{3} Z(\Omega) \times \frac{MVA}{(KV_L)^2}$$

$$Z_{pu} \propto (MVA)$$

$$\propto \frac{1}{(KV)^2}$$

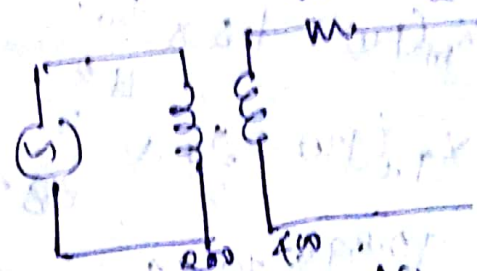
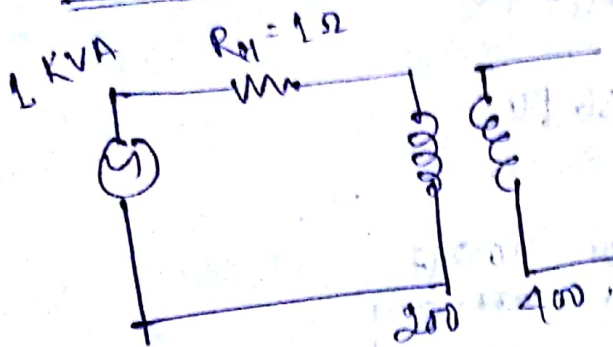
$$\frac{Z_{pu}(\text{new})}{Z_{pu}(\text{old})} = \frac{(MVA)_{\text{new}}}{(MVA)_{\text{old}}} \times \frac{(KV_b)_{\text{old}}}{(KV_b)_{\text{new}}}$$

$$Z_{pu}(\text{new}) = Z_{pu}(\text{old}) \times \frac{VA_{\text{new}}}{VA_{\text{old}}} \times \left(\frac{V_b \text{ old}^2}{V_b \text{ new}} \right)$$

$$R_{01} = 0.01 \times \left(\frac{N_2 R}{N_1} \right)$$

$$R_{01} = 1 \times \left(\frac{1}{2} \right)^2 = 1 \Omega$$

Y-connection



$$R_{01}(\text{pu}) = R_{01}(\Omega) \times \frac{VA}{V_b^2}$$

$$= 1 \times \frac{1000}{(200)^2}$$

$$= \frac{1}{40} \text{ pu}$$

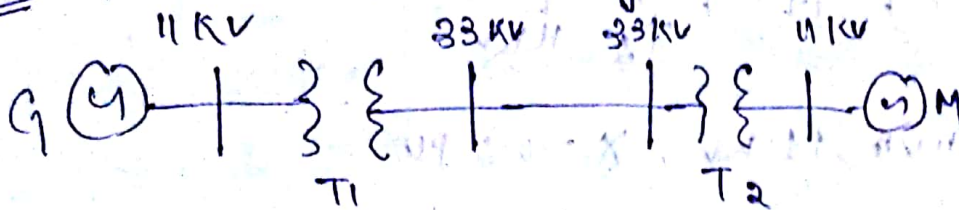
$$R_{01}(\text{pu}) = R_{01} \times \frac{VA}{V_b^2}$$

$$= 1 \times \frac{1000}{(400)^2}$$

$$= \frac{1}{40} \text{ pu}$$

Case-1 when reactance are given in Ω

Reactance Diagram



$$G = 30 \text{ MVA}, 11 \text{ KV}, X = 1.6 \Omega$$

$$T_1 = 15 \text{ MVA}, 11 \text{ KV}/33 \text{ KV}, X = 1.6 \Omega/\text{ph}$$

on HV side.

$$T_2 = 15 \text{ MVA}, 33 \text{ KV}/11 \text{ KV}, X = 1.7 \Omega/\text{ph}$$

LV side.

$$M = 25 \text{ MVA}, 11 \text{ KV}, X = 1.2 \Omega$$

$$TL = 20.5 \Omega/\text{ph}$$

draw reactance diagram by taking any base value.

30 MVA, 11 KV as base value at generating station

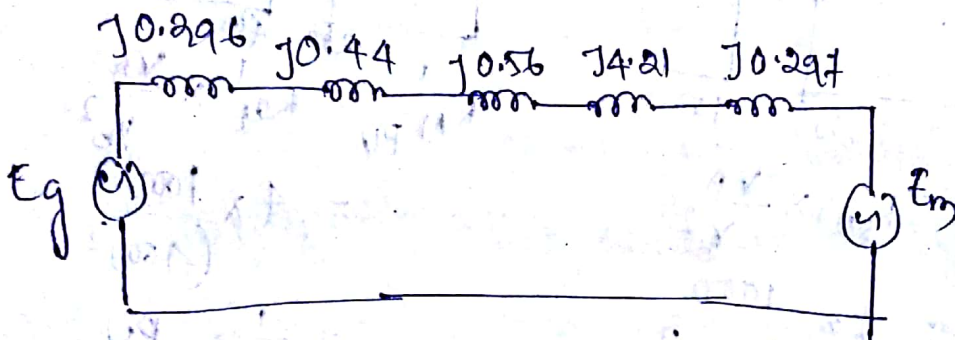
$$X_G(\text{pu}) = 1.6 \times \frac{30}{11^2} = 0.396 \text{ pu}$$

$$X_{T_1}(\text{pu}) = 16 \times \frac{30}{33^2} = 0.44 \text{ pu}$$

$$X_{T_2}(\text{pu}) = 17 \times \frac{90}{11^2} = 4.21 \text{ pu}$$

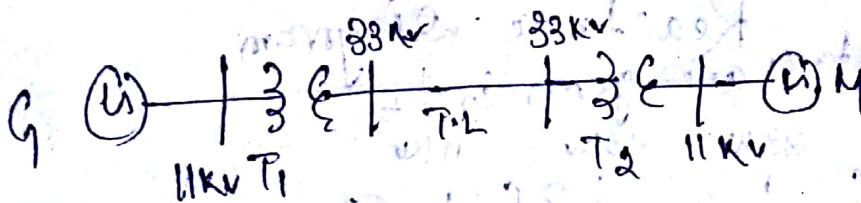
$$X_M(\text{pu}) = 1.2 \times \frac{30}{11^2} = 0.297 \text{ pu}$$

$$X_{T.L}(\text{pu}) = 20.5 \times \frac{90}{33^2} = 0.56 \text{ pu}$$



Case-II

When reactance are given in pu.



$$G = 30 \text{ MVA}, 11 \text{ KV}, X = 0.2 \text{ pu}$$

$$T_1 = 15 \text{ MVA}, 11 \text{ KV}/33 \text{ KV}, X = 0.1 \text{ pu}$$

$$T_2 = 15 \text{ MVA}, 33 \text{ KV}/11 \text{ KV}, X = 0.1 \text{ pu}$$

$$M = 25 \text{ MVA}, 11 \text{ KV}, X = 0.2 \text{ pu}$$

$$T.L = 20.5 \Omega/\text{ph}$$

Draw the reactance diagram by taking 100 MVA, 11 KV base voltage at G.

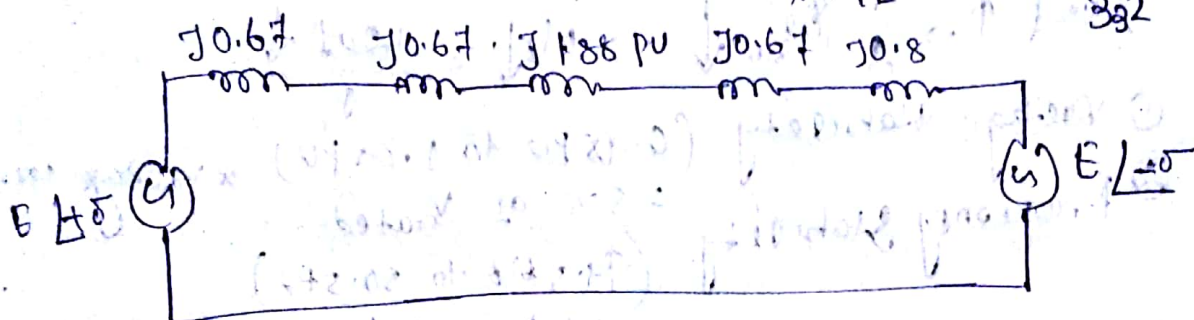
$$Z_{pu(new)} = Z_{pu(old)} \times \frac{(MVA)_{new}}{(MVA)_{old}} \times \left(\frac{V_{b,old}}{V_{b,new}} \right)^2$$

$$X_{G(new)} = 0.2 \times \frac{100}{30} \times \left(\frac{11}{11} \right)^2 = 0.67 \text{ pu}$$

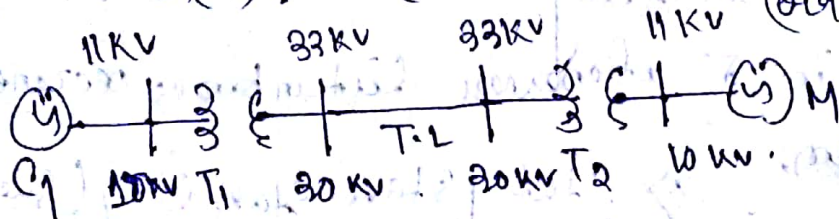
$$X_{T1(pu)} = 0.1 \times \frac{100}{15} \times \left(\frac{11}{11} \right)^2 = 0.67 \text{ pu} = X_{T2(pu)}_{new}$$

$$X_M = 0.2 \times \frac{100}{25} \times \left(\frac{11 \text{ kv}}{11 \text{ kv}} \right)^2 = 0.8 \text{ pu}$$

$$X_{TL} = 20.5 \times \frac{100}{302} = 1.88 \text{ pu}$$



Case-III where $V_b(new) \neq V_b(old)$.



$$G_1 = 30 \text{ mVA}, 11 \text{ KV}, X = 0.2 \text{ pu}$$

$$T_1 = 15 \text{ mVA}, 11 \text{ KV}/33 \text{ KV}, X = 0.1 \text{ pu}$$

$$T_2 = 15 \text{ mVA}, 33 \text{ KV}/11 \text{ KV}, X = 0.1 \text{ pu}$$

$$M = 25 \text{ mVA}, 11 \text{ KV}, X = 0.2 \text{ pu}$$

$$T.L = 20.5 \text{ } \Omega/\text{ph}$$

Draw reactance diagram by taking 100mVA, 10KV base voltage at G1.

$$X_{G(new)} = 0.2 \times \frac{100}{30} \times \left(\frac{11 \text{ KV}}{10} \right)^2 =$$

$$X_{T1(new)} = 0.1 \times \frac{100}{15} \times \left(\frac{11}{10} \right)^2 =$$

$$X_{T2(new)} = 0.1 \times \frac{100}{15} \times \left(\frac{11}{10} \right)^2 =$$

$$X_M(new) = 0.2 \times \frac{100}{25} \times \left(\frac{11}{10} \right)^2 =$$

$$X_{pu} = 20.5 \times \frac{100}{302}$$

Ch-3: POWER System Stability

→ The phenomenon of regaining the original operating point of a system subjected to marginal or conceivable disturbance within a time slot of 's' is considered as stability.

→ Stability can also be defined as ability of the system to transfer power continuously depending upon the intensity of the disturbance. The stability phenomenon is classified in three types.

Steady State Stability

The phenomenon of stability under slow & small changes is considered as steady state stability.

Ex: Small load change,

slight variations in excitation,

The parameter deviation in this phenomenon will not exceed by 5%.

→ The remedy are AVR and ALFC.

$$P = \frac{EV}{X} \sin \delta$$

power sent from AVR & ALFC

$$P = \frac{E_s \sin \delta}{X}$$

Transient Stability

The phenomenon of stability under large changes within a small time is considered transient stability.

Ex - sudden load change and faults.

→ The parameter deviation under this phenomenon is more than 5%.

→ The remedy or removal of Automatic Excitation Controller, (more than 5% less than 4% → AVR).

- i) use of parallel lines
 - ii) use of circuit breaker
- of bus setting with auto recloser

Dynamic Stability

The phenomenon of stability under parameter deviation within a narrow band of (2-3%) is considered as dynamic stability.

This phenomenon will never cause the system to become unstable or to exhibit oscillation.

→ These oscillation will deteriorate mechanical strength and system collapses physically.

These are called as low power frequency oscillation and are in the range of 0.2 Hz to 2.0 Hz.

This oscillation can be damped out by designing Power System Stabilizers.

These are pure reactive components. Raster is the equivalent in electrical system for the mechanical damper.

Assumption on Stability Studies

To carry stability studies on all the power system the following assumption must be considered:

① All the resistances including damper windings are assumed to be negligible to determine optimistic stability limit.

② Speed should always be synchronous so that frequency doesn't change. i.e. contraction or expansion of operating parameter is not allowed.

③ The synchronous machine should be modeled as single machine connected to infinite bus bar (BIB).

This modeling is also called as classical modeling.

④ All the loads are modeled as constant admittance model.

Mathematical Modeling.

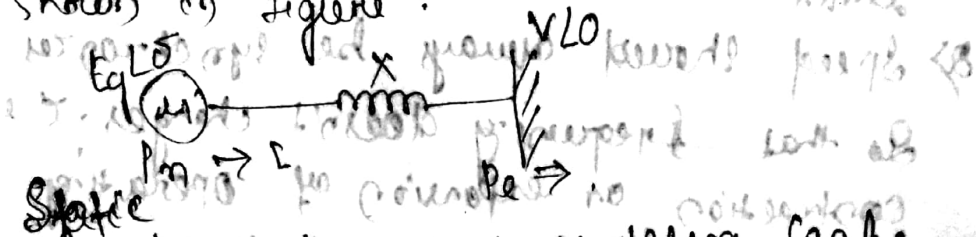
→ For the power system stability analysis the system should be properly modeled.

The power system is originating from generator there by system stability

is very much depend upon the stability. That's why more emphasis is given to synchronous machine

in power system stability analysis

The power system with synchronous machine is made as classical modeling as shown in figure.



→ The steady mathematical modeling can be considered as power angle characteristic as can be written as

$$P_e = \frac{E_g V}{X} \sin \delta$$

→ Now dynamic modeling should be obtained

The synchronous machine is supplied with mechanical torque of T_m

It generates equivalent electrical torque T_e under the condⁿ of equilibrium

$$T_m = T_e$$

→ If $T_m \neq T_e$

The machine is subjected to acceleration or deceleration.

If $T_m > T_e$ acceleration.

If $T_m < T_e$ deceleration.

There by the acceleration torque can be developed which is

the acceleration torque

$$T_a = J \frac{d^2\omega_m}{dt^2} + b \frac{d\omega_m}{dt} + T_e$$

As per the assumption taken $b=0, T_e=0$

$$T_a = J \frac{d^2\omega_m}{dt^2} = T_m - T_e$$

$$\omega_{max} \times J \frac{d^2\omega_m}{dt^2} = T_a = (T_m - T_e) \times \omega_{max}$$

$$J \omega_{max} \frac{d^2\omega_m}{dt^2} = P_a = P_m - P_e$$

where $\omega_{max} = (\frac{2}{p}) \omega_s$

$$\omega_m = (\frac{2}{p}) \omega_e$$

$$J \omega_s (\frac{2}{p})^2 \frac{d^2\omega_e}{dt^2} = P_a = P_m - P_e$$

$$J \omega_s (\frac{2}{p})^2 \frac{d^2\omega_e}{dt^2} = P_a = P_m - P_e$$

where $\omega_s = \frac{2\pi f}{p}$

$$\boxed{m \frac{d^2\omega_e}{dt^2} = P_a = P_m - P_e}$$

$$\omega_e = \frac{2\pi f}{p}$$

This eqn is considered as dynamic model of the system, as the rate of change of mathematical model is constant.

This eqn is based on a governing equation. The terminology is can be defined as

"The equation which describes the relative position of rotor w.r.t. stator magnetic field as a function of time."

It is the mathematical model that estimate how much fueling is required to maintain a steady state control force. This eqn is also called as motion differential eqn. Machine dynamic eqn.

Dynamics of M

Kinetic energy absorbed by rotating machine fundamentally can be expressed as

$$K.E. = \frac{1}{2} J \omega_m^2$$

Regarding the synchronous machine, with constant speed $\omega_m = \omega_s$.

The kinetic energy is defined as

$$K.E. = \frac{1}{2} J \omega_s^2$$

$$= \frac{1}{2} J \left(\frac{2\pi}{60} \right)^2 \omega_s^2$$

$$G_H = \frac{1}{2} m v^2$$

$$= \frac{1}{2} \times m \times \omega^2 r^2$$

$$\Rightarrow m = \frac{G_H}{\frac{1}{2} \omega^2 r^2} \text{ kg-sec}^2/\text{m}^2/\text{rad}^2$$

m can also be called as inertia of constant velocity difference.

Multimachine Modeling

The power system is consisting of more than one machines for the purpose of analysis it should be modeled. For the sake of mathematical explanation consider two machines.

The consideration of two w/o the modeling is carried with the consideration of

i) The M/c swinging together

ii) The M/c don't swing together.

Machines Swinging together

In this case the motor part of both the machines is same.

The total accelerating power is the summation of individual accelerating power.

$$P_a = P_{a1} + P_{a2}$$

$$m \frac{d\delta}{dt} = m_1 \frac{d\delta_1}{dt} + m_2 \frac{d\delta_2}{dt}$$

$$\text{As } \delta_1 = \delta_2 = \delta$$

$$M \frac{d\delta}{dt} = (m_1 + m_2) \frac{d\delta}{dt}$$

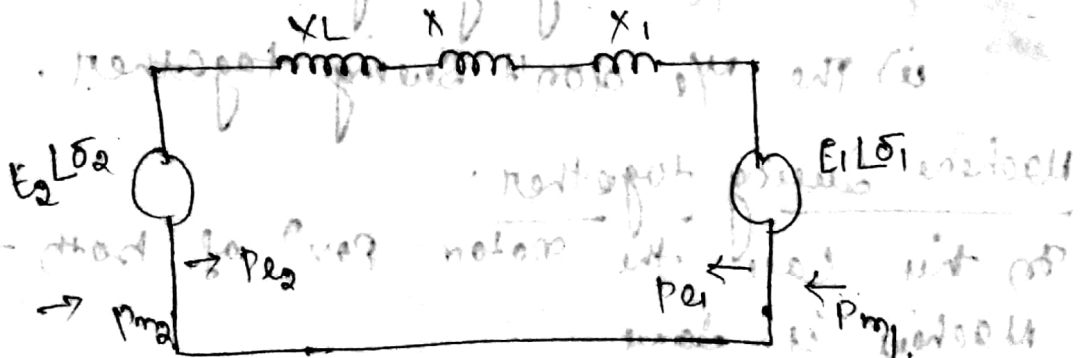
$$\Rightarrow M_{eq} = m_1 + m_2$$

The machine don't swing together. In this case the angular posⁿ of both the rotor are different.

One rotor may be advanced than other. There by the machine which is having more δ

can be subjected for generating action.

The machine which is having lesser load angle (δ) is subjected for motoring angle.



$$P_{m2} = -P_{m1} = P_m$$

$$P_{e2} = -P_{e1} = P_e$$

The net load angle can be consider as

$$\delta = \delta_2 - \delta_1$$

$$\frac{d\delta}{dt} = \frac{d\delta_2}{dt} - \frac{d\delta_1}{dt}$$

$$\frac{d^2\theta}{dt^2} = \frac{P_m - P_e}{m_1 m_2} \quad \left[\begin{array}{l} m \frac{d^2\theta}{dt^2} = P_m - P_e \\ \frac{d^2\theta}{dt^2} = \frac{P_m - P_e}{m} \end{array} \right]$$

$$\frac{d^2\theta}{dt^2} = \frac{m_1 m_2 (m_1 \theta_1 + m_2 \theta_2 - m_1 \theta_2 + m_2 \theta_1) - m_1 m_2 \theta}{m_1 m_2}$$

$$\Rightarrow \frac{d^2\theta}{dt^2} = \left(\frac{m_1 + m_2}{m_1 m_2} \right) (P_m - P_e)$$

$$\Rightarrow \left(\frac{m_1 m_2}{m_1 + m_2} \right) \times \frac{d^2\theta}{dt^2} = P_m - P_e$$

$$M_{eq} \cdot \frac{d^2\theta}{dt^2} = P_m - P_e$$

$$M_{eq} = \frac{m_1 m_2}{m_1 + m_2}$$

Problem

A 3φ Star connected 120 MVA alternator takes the mechanical input of 480 MW. Find the value of H.

input = KE = QH = 480

$$H = \frac{480}{120} = 4 \text{ MJ/MVA}$$

$$\frac{Q}{H} = \frac{480}{3.1416 \times 120} = 9.06 \text{ MJ-sec/rad}$$

$$H = \frac{Q}{\omega} = \frac{480}{3.1416 \times 120}$$

Q1) A pool of power system
 200 mVA alternator each has
 constant of 6 mJ/mVA and
 alternator each of inertia
 constant 4 mJ/mVA
 Find the Equivalent inertia
 constant referred
 to 1000 mVA.

Solⁿ
 In this case the total energy can be
 obtained as

$$\begin{aligned} \sum G.H &= 4 \times 200 \times 6 + 5 \times 200 \times 4 \\ &= 1200 + 4000 \\ &= 11200 \text{ mJ} \end{aligned}$$

This total energy should be equated to

$$\begin{aligned} G_{\text{eq}} H_{\text{eq}} &= 11200 \\ H_{\text{eq}} &= \frac{11200}{1000} = 11.2 \text{ mJ/mVA} \end{aligned}$$

Q2) 2 synchronous machine are combine to
 operate with inertia constant of
 $m_1 = 10$, $m_2 = 15$. Find the
 equivalent inertia constant when both
 the machine swing together
 ii) ~~do not~~ swing together

Solⁿ
 i) Both machine swing together
 $= m_1 + m_2 = 10 + 15 = 25$

ii) Both the machine don't swing together

$$\frac{m_1 m_2}{m_1 + m_2} = \frac{10 \times 15}{10 + 15} = 6$$

Steady State Stability - Analysis

The phenomenon of stability under slow & small changes is considered as steady state stability.

Under this stability changes the system stability can be determined by making proper analysis.

→ Let us consider a system is maintaining equilibrium in such a way that
$$P_m = P_{e0} \text{ at } \delta_0$$

If the load is increased by ' ΔP ' the electrical power becomes as

$$P_e = P_{e0} + \Delta P$$

The load angle also changes correspondingly to such an extent that

$$\delta_1 = \delta_0 + \Delta \delta$$

The system stability corresponding to this new load angle should be determined:

$$P = \frac{EV}{X} \cos \delta$$

$$\Delta P = \frac{d}{d\delta} \left(\frac{EV}{X} \cos \delta \right) \Delta \delta$$

$$= \frac{dP_e}{d\delta} \Delta \delta$$

$$m \frac{d^2 \delta}{dt^2} = P_m - P_e$$

$$m \frac{d^2 (\delta_0 + \Delta \delta)}{dt^2} = P_m - (P_{e0} + \Delta P_e)$$

$$m \frac{d^2 \Delta \delta}{dt^2} = -\Delta P_e$$


$$m \frac{d^2 \Delta \delta}{dt^2} + \frac{\partial P_e}{\partial \delta} \Delta \delta = 0$$

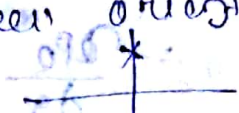
$$\left(\frac{d^2}{dt^2} + \frac{\partial P_e}{\partial \delta} \right) \Delta \delta = 0$$

$$\Delta \delta \text{ is proportional to } \sqrt{\frac{\partial P_e}{\partial \delta}}$$

The orientation of stability of the system depends upon absolute values of m and $\frac{\partial p_e}{\partial \delta}$.

m can have always the value but $\frac{\partial p_e}{\partial \delta}$ can have both the or -ve value.

1) If $\frac{\partial p_e}{\partial \delta} < 0$, roots are real and different. They orient as  then the system becomes unstable.

2) If $\frac{\partial p_e}{\partial \delta} > 0$, the roots are imaginary and conjugate. They orient as .

Originally this can be stated as marginal stable condition but in this case we can be considered as stable condition because this system could be able to manage up to marginal stable condition. even though the natural damping is seized?

For every practical system we definitely have natural damping. This amount of natural damping is sufficient to give the system stability there by the consideration of

$\frac{\partial P_e}{\partial \delta} > 0$ is taken as condⁿ for stability.

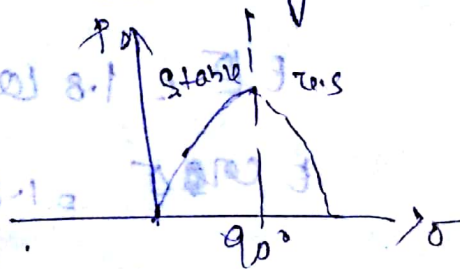
$$\frac{E_V}{X} \cos \delta > 0.$$

$$\cos \delta > 0.$$

$$\left| \delta < 90^\circ \right|.$$

From this it can be stated that under steady state condition load can be increased until the load angle increases to go to retain steady state stability until the reach of 90° load angle 90° , what ever the load increases that is on the

steady margin. on this phenomenon the stability concern is determined by angle therefore this stability phenomenon is also called as angle stability.



Steady State Stability Limit

It is the max^m power that can be delivered or transferred by machine or system without losing synchronism.

It can be estimated as

$$P = \frac{E_V}{X} \sin \delta$$

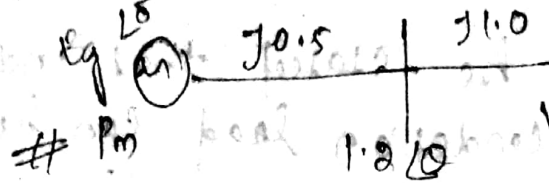
at steady state condition $\delta = 90^\circ$

$$P_{st} = \frac{E_V}{X} \sin 90^\circ = \frac{E_V}{X}$$

$$P_{st} = \frac{E_V}{X} = P_{e \max}$$

Problem

Find the steady state stability limit of the configuration given below. All the values are specified in p.u.



Solⁿ

$$P_{cs} = \frac{E_v}{X_{eq}}$$

$$E \cos \delta = 1.2 \cos \delta + (1.0) 0.5$$

$$E \sin \delta = \frac{1.2 \sin \delta - 1.0 \sin \delta}{1.0}$$

$$E \cos \delta = \frac{1.2 \cos \delta + 1.2 \cos \delta - 1.0 \cos \delta}{1.0} \times 0.5$$

$$E \cos \delta = 1.8 \cos \delta - 0.5$$

$$E \cos \delta = 1.8 \cos \delta - 0.5$$

for 2 eq condition $\delta = 90^\circ$

$$E \cos 90^\circ = 0 = 1.8 \cos 90^\circ - 0.5$$

$$0 = \cos^{-1} \left(\frac{0.5}{1.8} \right) = 73.8^\circ$$

$$E \sin 90^\circ = 0 = 1.8 \sin 73.8^\circ - 0.5$$

$$= 1.73$$

$$P_{cs} = \frac{1.0 \times 1.73}{1.5} = 1.15 \text{ p.u.}$$

Transient Stability
The phenomenon of stability under large changes with small time is considered as transient stability.

System stability under transient condition can be determined by two techniques

- > Energy balancing method
 - > This is invented by Liapunov & Papov
 - > Every system under disturbing condition will lose some part of kinetic energy
 - > under this condition, it can be said as stable one if it could be able to develop the P.E. of more or equal to the kinetic energy loss.

> By Solving Machine differential Equation
under the disturbed condition as per the data type and system configuration, the machine differential eqn is formed as

$$m \frac{d^2 \delta}{dt^2} = P_a - P_m - P_e$$

This machine differential equation is solved in terms of δ

- Commonly used methods are
- Step by step method
 - Modified Euler's method
 - Runge-Kutta 4th order method

in any method the algorithm should be designed to get the solution

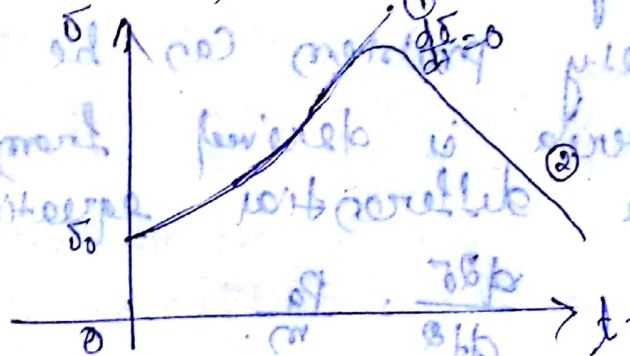
$$\frac{d^2\delta}{dt^2} = \frac{Pa}{m}$$

$$\Rightarrow \frac{d\delta}{dt} = \frac{Pa}{m} t + k \quad \text{at } t=0, \delta = \delta_0$$

$$\Rightarrow \delta = \frac{Pa}{2m} t^2 + k_1 t + k_2 \quad \text{and } k_1 = \dot{\delta}_0$$

$$\delta = \delta_0 + \frac{Pa}{2m} t^2$$

with this soln a graph is plotted which shows that δ increases continuously as shown in figure.



with increase in time δ can also increase continuously as shown in figure. Under this case system is considered to be unstable.

This is possible when Pa assumes to be constant force it is not constant again it is function of δ , and also on the -ve side.

There by incrementation of δ up to certain extent may be possible from then onwards it decreases as shown in figure under this case.

we can say system is coming towards region of convergence of stability. It has initiated from the first

$$\frac{d\sigma}{dt} = 0$$

This condition gives us conditional stability in the studied.

Equal Area Criterion

It is a mathematical criteria developed for the determination of stability under transient condition.

By making use of this criteria transient stability problem can be analysed.

The criteria is derived from fundamental machine differential equation.

$$\frac{d\sigma}{dt} = \frac{P_a}{m}$$

Multiplying both sides with $\frac{d\sigma}{dt}$

$$\frac{d\sigma}{dt} \frac{d\sigma}{dt} = \frac{P_a}{m} \frac{d\sigma}{dt}$$

Integrate both sides of the equation

$$\int \frac{d\sigma}{dt} \frac{d\sigma}{dt} = \int \frac{P_a}{m} d\sigma$$

For stability $\frac{d\sigma}{dt} = 0$

$$\int \frac{P_a}{m} d\sigma = 0$$

Therefore $\int P_a d\sigma = 0$

This eqⁿ mathematically is not correct because the objective of sign convention

not give definite value. There by it should be considered as

$\int \delta \dot{\omega} dt = 0$
This integral specifies area bounded by a curve tracing defined between limits $\delta \omega_1$ & $\delta \omega_2$ is equal to zero. i.e. The curve tracing binding two equal & opposite area.

→ The +ve area bounded by P- δ curve is called as acceleration A_1 .

→ The -ve area bounded by P- δ curve is called as acceleration A_2 .

For the system to be stable

$$A_1 = A_2$$

This eq is called as equal Area criterion.

NOTE

✓ On the operation of power system acceleration is always considered as unstable phenomenon. deceleration is considered as stable phenomenon.

Conditional Stability is considered as

$A_1 = A_2$ under critical condition only. where as the is absolute

condition for stability is

This means $\Delta \dot{\theta} > 0$.
The means decelerating area is more than accelerating area. Under this condition system couldn't be able to get one more acceleration.

→ 1st Swing stability is guaranteed achieved.

Analysis:

The system should be analysed against transient condition. Let say in the machine by two consideration.

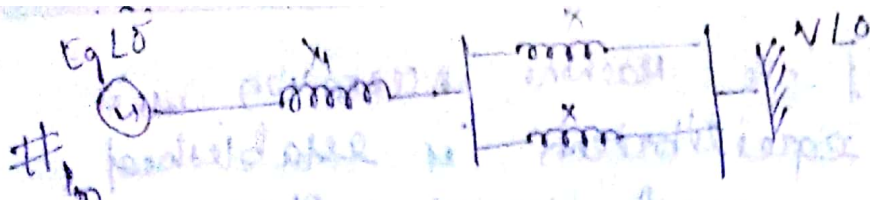
1) Keeping the electrical power output constant, sudden rise in mechanical input.

2) Keeping the mechanical input constant increment in electrical power.

→ Under this two condition the accelerating power is $F > 0$, and the machine is subjected to acceleration.

→ Under this condition their might be determination condition of stability through analysis.

→ For this the following system can be taken as test system.

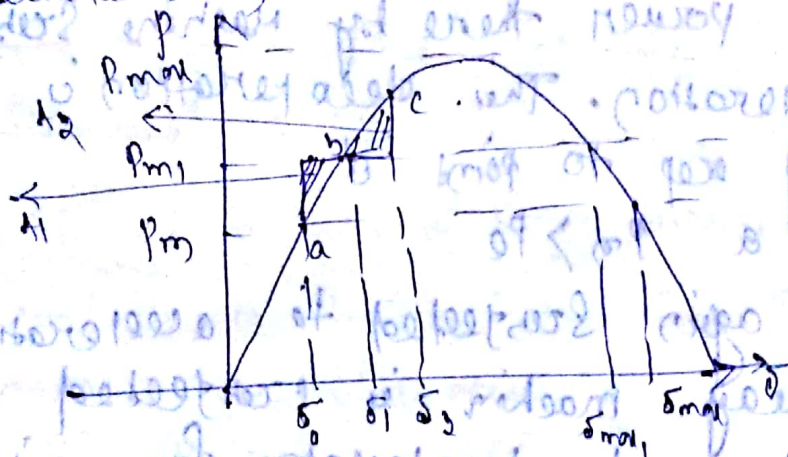


$$\delta_e = \frac{E_g V}{X_s + X/2} \sin \delta$$

$$\frac{m d\delta \omega}{d t} = P_a = P_m - P_e$$

→ when the mechanical input is suddenly raised,

under this case machine definitely subjected acceleration.



→ Initially system is assumed to be under equilibrium condition here by machine operator at operating point 'a'.

The corresponding load angle is δ_0 .

→ If mechanical input suddenly rises to because of excess kinetic energy rotor starts advancing and

load angle increases. Machine is subjected to acceleration → with this increase in

load angle - electrical power also increases continuously and the new operating point will be 'B'.

→ At the if the machine acceleration is stopped equilibrium is established, we can come out of transient phenomenon.

→ Also due to lack of damping the swing may not be considered as control swing and can be continued to other equilibrium point.

→ Let us consider a point c' at which electrical power is more than mechanical power there by machine subjected to deceleration. This deceleration is continued up to point d' .

→ At point a $P_m > P_e$ machine again subjected to acceleration.

→ In this way machine is subjected to acceleration and deceleration from a to c' and c' to a .

→ Under this condition machine can be set as stable one if it releases the energy when it is at deceleration and it grasped at acceleration.

→ The two energy calculated in terms of two areas

$$A_1 = \int_{\delta_1}^{\delta_2} (P_m - P_{e \max} \sin \delta) d\delta$$

$$A_2 = \int_{\delta_2}^{\delta_1} (P_{e \max} \sin \delta - P_m) d\delta$$

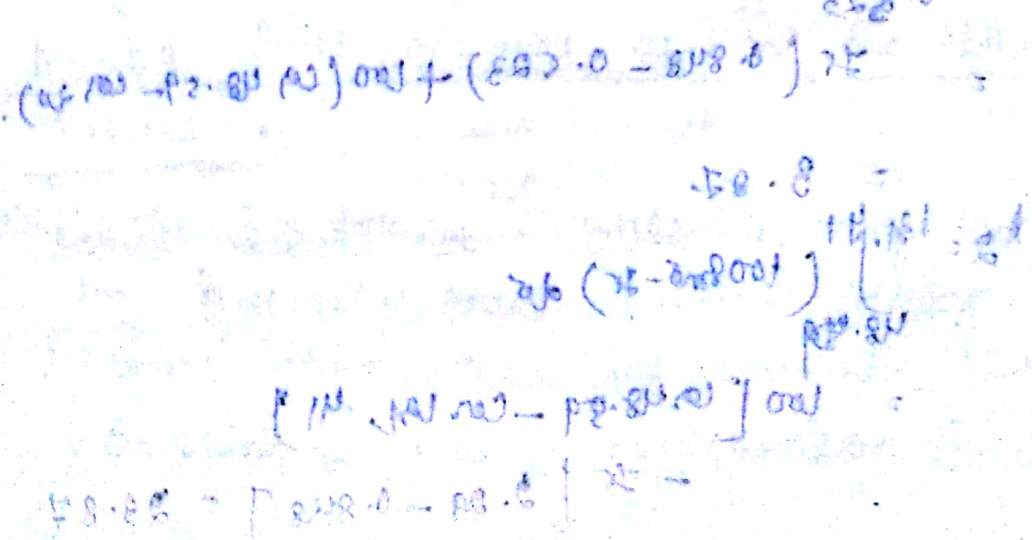
System is stable if $\phi_m = 0$
→ as per from the figure it can be said that at this point $\phi_m = 0$.

There by the system can be said to be stable one from this point onwards.

→ The point can be extended upto the angular posn of ϕ_m , which is greater than 90° .

→ From this it can be said that for the operating lead angle beyond 90° there is transient stability only.

→ For this step rise of PM system is found to be stable one. If we further increase the step rise we observe to be orienting towards right and ϕ_m orienting towards left. Under this condition we can have max step rise in such a way that ϕ_m coincide with ϕ_{max} .



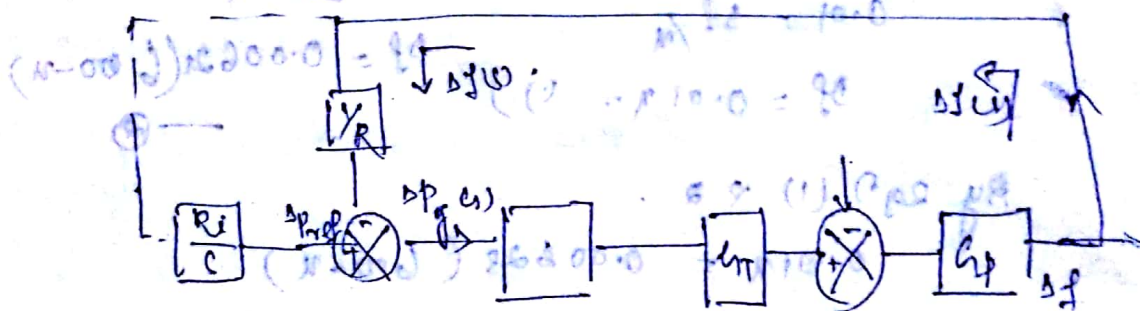
Secondary ALFC

A load frequency controller is considered as Primary ALFC. Control action is equal to zero to its controller.

The controller adopted here is proportional controller which always reduces the error.

To make error and static deviation to be zero, integral controller should be adopted.

The Primary ALFC with integral controller can be considered as secondary ALFC.



$$\Delta f(s) = \frac{K_p}{s} \frac{\Delta f_R(s)}{(1+STP) \left(\frac{K_p}{s} + K \right)}$$

$$\Delta f(s) = \lim_{s \rightarrow 0} s \Delta f(s)$$

$$\Delta f_0 = \lim_{s \rightarrow 0} \Delta f(s) \times 10.0 = 2.0$$

From this it can be proved that the static deviation with integral controller is equal to zero.

The system is called as single area system.

System

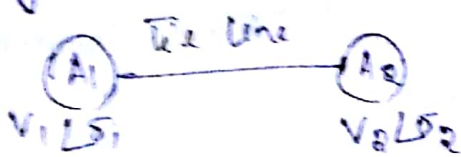
In the power system, sometimes it is necessary to connect two area for the

Low of reliability of the power transfer.
 The two areas are interconnected with a
 connection is called tie-line.



Tie-line modeling

Tie-line is the interconnection of the two power systems held at different operating frequency it is sought to transfer the power from A_1 to A_2 . In this case we have to maintain the two areas are strong and tie-line is weak. It is preferred



$$P_{12} = \frac{V_1 V_2}{X} \sin(\delta_1 - \delta_2)$$

$$\frac{dP_{12}}{dt} = \frac{V_1 V_2}{X} \cos(\delta_1 - \delta_2) (\omega_1 - \omega_2)$$

$$P_{12} = T \sin(\delta_1 - \delta_2)$$

$$\omega = \frac{d\delta}{dt}$$

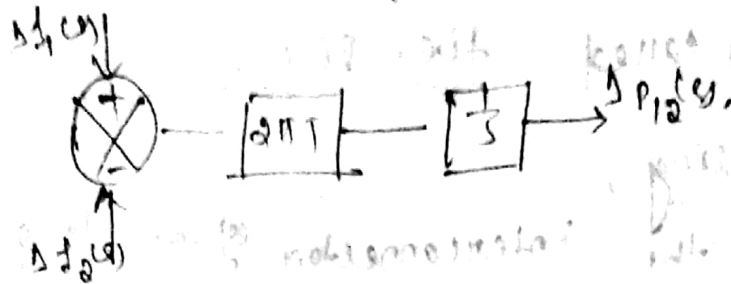
$$P_{12} = T \sin(\delta_1 - \delta_2)$$

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$P_{12} + P_{21} = P_{12}$
 $P_{12} + P_{21} = P_{12}$

$$\Delta P_{12}(s) = \frac{sT}{s} (\Delta f_1(s) - \Delta f_2(s))$$



Two Area Systems

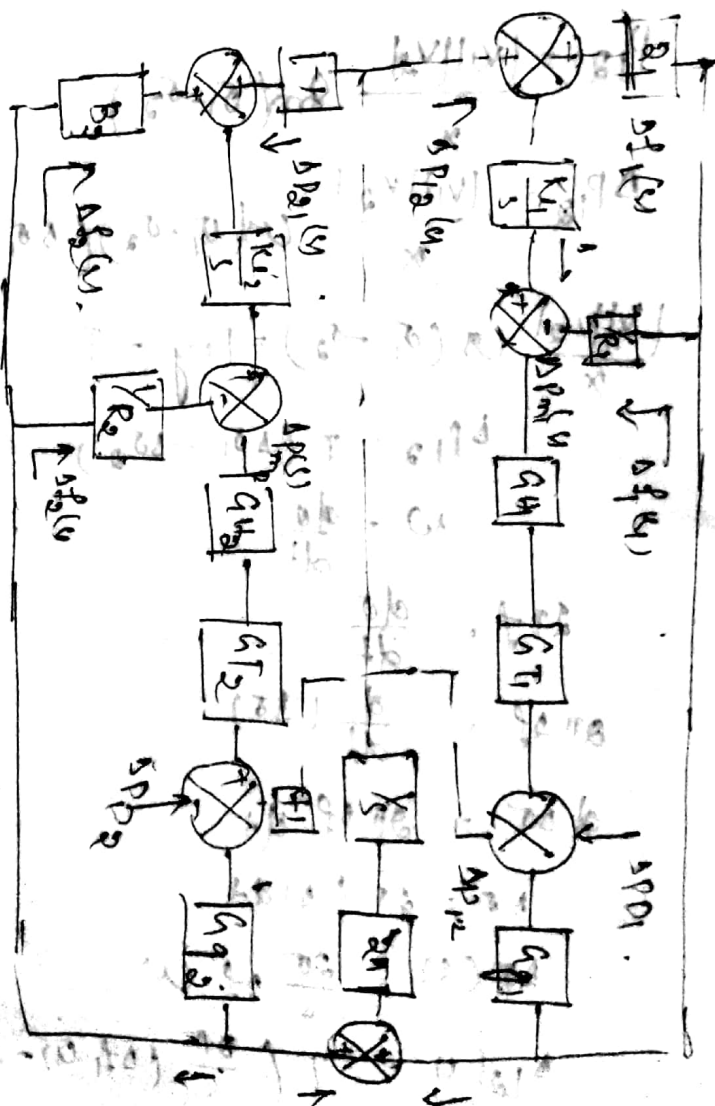
The tie-line is used for the controlling of two area as interconnected and its modeling is as shown in figure.

Two area are inter

ans -

$$ACE_1 = \Delta P_{12} + B_1 \Delta f_1$$

$$ACE_2 = \Delta P_{21} + B_2 \Delta f_2$$



The best efficiency of B parameter.

power factor $\cos \phi_1$ is $\cos \phi_1$ and $\cos \phi_2$ is $\cos \phi_2$

$$P_1 = \frac{V_1^2}{R_1} + X_{R1}$$

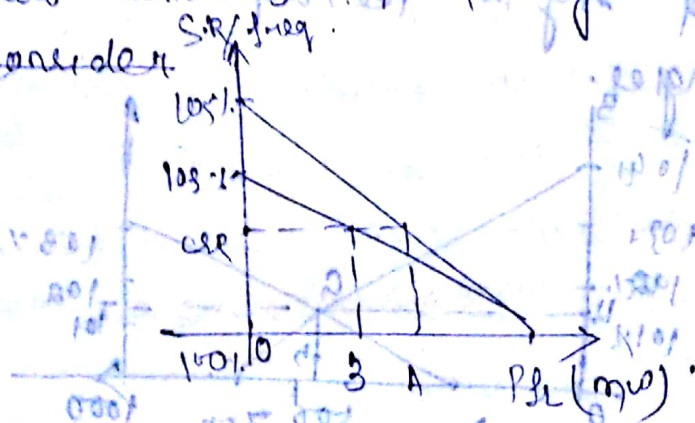
Load Sharing

The load sharing depends upon drooping characteristics

The synchronous machines connected in parallel need to operate constant voltage

the drooping were defined on frequency (speed)

Let us consider



If the two generators are connected in parallel, they will operate at a common speed regulation relation which is less than 100%.

The intersection at CER means speed regulation characteristics as shown in fig.

From this it can be stated that 'O.A' is the load shared by 1st machine and 'O.B' is the load shared by 2nd machine.

Relatively $0.1 > 0.3$.
 From this it can be concluded that
 the machine which is having slow speed
relation (+w) with change in load.

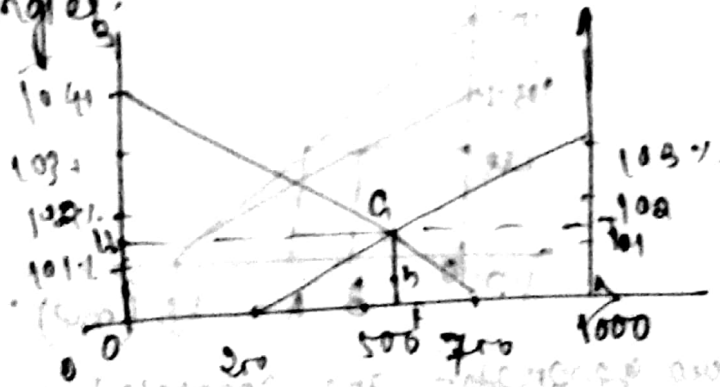
Problem

Two 200 Ampere Alternators are connected
 in parallel to share a common load of
 1000 amp. The speed relation characteristics

are
 $104\% \rightarrow 100\%$
 $103\% \rightarrow 100\%$ from no load to

full load. Find the load sharing at
 and operating frequency.

This data should be formulated and can be
 solved by the technique of model as
 arranged.



At 700 A, the load is shared as follows:
 $\frac{104 - 102.5}{104 - 101} \times 1000 = 500$ A
 $\frac{103 - 102.5}{103 - 100} \times 1000 = 200$ A

$\rightarrow 700 - 127.5$

Two A.C. D.C. are similar

$$\frac{dP}{dE} = \frac{dP}{dE}$$

$$P_1 = 750 \left(\frac{5-h}{5} \right)$$

$$= 750 - 200h$$

Total load = 1000 = $P_1 + P_2$

$$1000 = 750 - 187.5h + 750 - 200h$$

$$h = \frac{500}{437.5} = 1.142$$

Load shared in 1/c 1 = $750 - 187.5 \times 1.142 = 535.8$

1/c 2 = $750 - 200 \times 1.142 = 464.2$

$$f_2 = f_1 \left(1 + \frac{h}{100} \right)$$

$$= 50 \left(1 + \frac{1.142}{100} \right) = 50.57 \text{ Hz}$$

This set can handle near load is betw 1465 to 1485. because the speed relation of both the generator is not same so their might be a small difference by which capacity is less than the rated capacity.

This set can operate at a load of 187.5 MW where generation from to one generator is only possible.

but $f_2 = 50.57$ Hz
 but $f_1 = 50$ Hz
 but $f_2 = 50.57$ Hz